

```
%-----
%      Rotary Motion Inverted Pendulum System for Quanser JP
%
%      This program is nonlinear system
%
%      Motor input -24V <-----> +24V
%      振り上げのみのプログラム+LQGコントローラ
%
%      Notice!
%      call swingup12_eq12.m
%      call swingup11_LQG(simulink program)
%
%      2006. 7. 26           produced by kizawa
%-----
```

```
clc
clear all
format long e

global MD

MD=zeros(2, 2);

%----setup---%
sampling=0.001;%sampling time of simulation
kaisu=1000;
timespan=sampling*kaisu;% simulation time
%--- initial state ---
th0=0.00; %angle of arm(rad)
th1= -1*pi; %angle of pendulum(rad)          -3.14:真下
d_th0=0.0; %angle velocity of arm(rad/s)
d_th1=0.0; %angle velocity of pendulum(rad/s)

x0=[th0 th1 d_th0 d_th1]; %initial condition
%manipulated variable
Vin=0.0; %[Volt]
u2=0.0;%%%%%%%%%%%%

%---mechanical parameter---
mp = 0.067; %Mass of Pendulum[Kg]半分0.066
ma = 0.057; %Mass of Arm[Kg]
r = 0.137; %length of Arm[m]
ra= 0.0685; %length to the center of gravity of arm[m]
lp= 0.155; %length to the center of gravity of pendulum[m]0.167
Lp=0.317; %length of pendulum
la=2.67e-4; %arm inertia moment of the circumference of the center of gravity1.2e-3
lp=0.971e-3; %pendulum inertia moment of the circumference of the center of gravity
Jm=2.19e-6; %inertia moment of motor[kgm^2]
Ra = 5.7; %armature resistance[ORM]
Km = 3.038e-2; %Motor torque constant0.023[Nm/A]
Ke =3.06e-2; %[Vrad/s]
ca=2.919e-3; %[Nms/rad]armエンコーダ0.2016摩擦を小さく0.04
cp=1.671e-4; %[Nms]pendulumエンコーダ 1.899e-4
n = 29.47; %gear ratio 79.24
g = 9.8; %[m/s^2] acceleration of gravity
```

Jb=1.2e-4;%[kgm^2]アームと取り付け部を含むInertia;1.15e-4

$$Kt \leftarrow n * Km / Ra;$$

$$cti=ca + n^2 * Ke * Km / Ra;$$

%----- Non linear system

%---definition---

%アーム角度	振子角度	アーム角速度	振子角速度	アーム角加速度	振子角加速度	電圧
%---y (1, 1)	y (2, 1)	y (3, 1)	y (4, 1)			u
%		dy (1, 1)	dy (2, 1)	dy (3, 1)	dy (4, 1)	
%						

%---Y(1, 1) Y(2, 1) Y(3, 1) Y(4, 1) PDY(1, 2) PDY(2, 1)

%アーティクル角度 振子角度 アーティクル角速度 振子角速度 アーティクル角加速度 振子角加速度

10%

%-----Liner System ↵

10

%-----Liner System ↵

$$E = \frac{1}{2} m v^2 + \frac{1}{2} I \omega^2 = \frac{1}{2} m r^2 \dot{\theta}^2 + \frac{1}{2} I_{\perp} \dot{\phi}^2$$

F=

$r = \frac{m}{\rho}$

$$A1 = [-ca - n^2 * Ke * Km / Ra \quad 0 \\ 0 \quad -cp];$$

$$A2 = [0 \quad 0 \\ 0 \quad mp*g*|p|];$$

$$B_1 = [n * K_m / Ra \quad 0]'; \% 間違い$$

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A=[zeros(2, 2) eye(2, 2)
    inv(E)*A2 inv(E)*A1];
```

```
B=[zeros(2, 1)  
    inv(F)*B1];
```

$$C = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix};$$

```
D=zeros(2,1);
```

%----- LOG/ITR ↵

Controller _____ %

%---IQ Controller---%

$$Q = [1000 \quad 0 \quad 0 \quad 0 \\ 0 \quad 10.0 \quad 0 \quad 0 \\ 0 \quad 0 \quad 10 \quad 0 \\ 0 \quad 0 \quad 0 \quad 3]; \quad \%3$$

$$R = [0, 4]; \quad \%0.0025 \quad 0.1$$

aa=1e3; %1e3

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PP=care(A, B, Q, R); %Riccati equation
disp('FeedbackGain')
Fgain=inv(R)*B'*PP %feedback gain [V/rad]
disp('Eigen value A-BF')
eigen_Af=eig(A-B*Fgain)

%K=K/180*3.14; %feedback gain [V/degree]
SS=care(A', C', qq * B * B'); %Riccati equation
disp('ObseberGain')
Kalgain=SS * C' %kalman felter gain
disp('Eigen value A-Kalgain * C')
eigen_Ak=eig(A-Kalgain*C)

Ak=A - Kalgain * C - B * Fgain;
Bk=Kalgain;
Ck= -Fgain;
Dk=[0 0];
disp('Controller Eigen value A - Kalgain * C -B * Fgain ')
eigen_Ak=eig(Ak)

disp('コントローラの固有値は')
eig(Ak)

%---閉ループ系 -----%
disp('閉ループ系の固有値は')
AG=[A B*Ck
     Bk*C Ak];
eig(AG)
%-----boundary condition---
ipu=1.0e-1; %0.1
delta=0.02; %0.04 0.03
Vmax=24;%24[V]10

%-----
disp('It is under calculation now!!')
ni=1;
tend=5000; %7000
while(ni <= tend)
    %x0(1,3)

%%%%% Jump to LQG/LTR Control for Liner System %%%%%%
%この条件も-360に収束するときは変更が必要、レギュレータも-360に収束することを検討する必
要がある
if ( ( x0(1,2) > -pi/6 ) && ( x0(1,2) < + pi/6 ) ) % -pi/6 この条件では厳しい
    break;
end
%%%%%



if ( -ipu <= x0(1,3) ) && ( x0(1,3)<=ipu )
    if x0(1,3) >= 0

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        Vin= Vmax * (+1);
elseif x0(1, 3) < 0
        Vin= Vmax * (-1);
end

else
    if x0(1, 3) >= 0.0

        u2=( cp*x0(1, 4)^2/x0(1, 3) ) + delta;
        Vin=( u2 + cti*x0(1, 3) )/Kti ;
        if Vin >=Vmax
            Vin=+Vmax;
        end
elseif x0(1, 3) < 0.0

        u2=( cp*x0(1, 4)^2/x0(1, 3) ) - delta;
        Vin=( u2 + cti*x0(1, 3) )/Kti ;
        if Vin <= -Vmax
            Vin= -Vmax;
        end
    end
end

%%%%%アームが回り過ぎないように%%%%%
if      ( x0(1, 1) > pi*0.5 )           %スイングを規制すればエネルギーが一端下がる0.7
    Vin=-Vmax;  %-1.3

end
if      ( x0(1, 1) < -pi*0.5 )
    Vin=+Vmax;  %+1.3

end
%%%%%
[Tn, Yn]=ode45 (@swingup_12_eq12, [0 sampling], x0, [], Vin, ca, cp, mp, r, lp, Ip, Jb, Ra, Km, Ke, Jm, ↵
n, g);

if ni==1
    YY(1, :)=Yn(1, 1:4);
    TT(1, :)=Tn(1, 1);
    VIN(1, :)=Vin;
    U2(1, :)=u2;
    %---motion enegy---%
    ENE(1, 1)=0.5*[Yn(1, 3)  Yn(1, 4) ] * MD * [Yn(1, 3)  Yn(1, 4)]' + mp*g*lp*cos( Yn(1, 2) ↵
);
    %-- Dot motion enegy ---%
    DENE(1, 1)=u2*Yn(1, 3) - cp*( Yn(1, 4) )^2;
else
    YY(ni, :)=Yn(size(Yn, 1), 1:4);
    TT(ni, 1)=sampling*ni;
    VIN(ni, 1)=Vin;
end

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U2(ni,:)=u2;
%---motion enegy---%
ENE(ni,1)=0.5*[Yn(1,3) Yn(1,4)] * MD * [Yn(1,3) Yn(1,4)]' + mp*g*lp*cos( Yn(1,2) ↵
);
%-- Dot motion enegy ---%
DENE(ni,1)=u2*Yn(1,3) - cp*( Yn(1,4) )^2;
end
%---- Up data-----
x0=Yn(size(Yn,1),1:4);
ni=ni+1;
end

ni

figure(1)
subplot(2,2,1)
plot(TT, 180/pi*YY(:,1));
axis([0 sampling*ni -360 +360])
xlabel('Time[s]');
ylabel('Angle of Arm[degree]')

subplot(2,2,2)
line_bottom=-180*ones(1,size(YY,1));
line_top0=zeros(1,size(YY,1));
line_top1=-360*ones(1,size(YY,1));
line_dome1=30*ones(1,size(YY,1));
line_dome2=-30*ones(1,size(YY,1));
line_dome3=-330*ones(1,size(YY,1));
line_dome4=-390*ones(1,size(YY,1));

plot(TT, 180/pi*YY(:,2),'-k', TT, line_bottom,'-r', TT, line_top0,'-r', TT, line_top1,'-r', .....
TT, line_dome1,'-b', TT, line_dome2,'-b', TT, line_dome3,'-b', TT, line_dome4,'-b');
axis([0 sampling*ni -450 +90])
xlabel('Time[s]');
ylabel('Angle of pendulum[degree]')
title('theta=0(deg), theta=-360(deg) ==>TOP')

subplot(2,2,3)
plot(TT,VIN);
axis([0 sampling*ni -25 +25])
xlabel('Time[s]');
ylabel('INPUT Moter Voltage [V]')

subplot(2,2,4)
plot(TT,U2);
%axis([0 sampling*ni -1 +2.0])
xlabel('Time[s]');
ylabel('Input u2')

figure(2)
subplot(2,2,1)
plot(TT,YY(:,3));
axis([0 sampling*ni -15 +15])
xlabel('Time[s]');
ylabel('Angular velocity[rad/s]')

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title('Angular velocity of Arm')

subplot(2, 2, 2)
plot(TT, YY(:, 4));
axis([0 sampling*ni -15 +15])
xlabel('Time[s]');
ylabel('Angular velocity[rad/s]')
title('Angular velocity of Pendulum')

subplot(2, 2, 3)
zeroline=zeros(1, size(YY, 1));
plot(TT, DENE, TT, zeroline, '-r');
axis([0 sampling*ni -1 +6])
xlabel('Time[s]');
ylabel('Dot Energy')
title('Dot Energy')

subplot(2, 2, 4)
plot(TT, ENE);
axis([0 sampling*ni -1 +3.0])
xlabel('Time[s]');
ylabel('Energy [J]')

%%%%% LQG/LTR Control for Liner System %%%%%%
%==== Call Simulink=====
% $x_0(1, 3)=0;$ 
% $x_0(1, 4)=0;$  %これがゼロであれば制御が容易
XX0=x0; %initial value for simulink 速度を持ってこの領域に入ることに注意
modelname='swingup12_LQG'; %change
FixedStep=sampling;
OPTIONS=SIMSET('FixedStep', sampling); %sampling set for simulink
timespan=[ni*sampling tend*sampling]; %timespan=[tStart tFinal]
tt=sim(modelname, timespan, OPTIONS);
%=====
%---motion enegy---
% $E_{NP}(1, 1)=0.5*[Y_n(1, 3) \ Y_n(1, 4)] * MD * [Y_n(1, 3) \ Y_n(1, 4)]' + mp*g*lp*cos(Y_n(1, 2));$ 
%=====

YY1=vertcat( YY(:, 1), TH0); %アーム角度 ベクトル垂直連結
TT=vertcat( TT, tt); %振子角度 ベクトル垂直連結
YYY2=vertcat( YY(:, 2), TH1); %アーム角速度 ベクトル垂直連結
YYY3=vertcat( YY(:, 3), THV0); %振子角速度 ベクトル垂直連結
YYY4=vertcat( YY(:, 4), THV1);

%---transient response for nonliner system and liner system---
figure(3)
subplot(2, 2, 1)
plot(tt, [180/pi*TH0 180/pi*TH1])
ax1=findobj(gca, 'Type', 'Axes', 'Visible', 'on');
a11=findobj(gca, 'Type', 'Line', 'Visible', 'on');
set(a11(1), 'LineStyle', '-')
set(a11(2), 'LineStyle', ':')
axis([ni*sampling tend*sampling -100 100])
set(ax1, 'fontsize', 12) %legend
%title(' Transient Response with initial condition [th0 th1]')

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```
xlabel('TIME(s)', 'FontSize', 12)
ylabel('Angle(degree)', 'FontSize', 12)
legend('Arm', 'Pendulum', 1)
grid on

figure(3)
subplot(2, 2, 2)
plot(TTT, YYY3(:, 1));
axis([ni*sampling sampling*tend -15 +15])
xlabel('Time[s]');
ylabel('Velocity of Arm[rad/s]')
grid on

figure(3)
subplot(2, 2, 3)
plot(TTT, YYY4(:, 1));
axis([ni*sampling sampling*tend -15 +15])
xlabel('Time[s]');
ylabel('Velocity of Pendulum[rad/s]')
grid on

figure(4)
subplot(2, 2, 1)
plot(TTT, 180/pi*YYY1(:, 1));
axis([0 sampling*tend -200 +200])
xlabel('Time[s]');
ylabel('Angle of Arm[degree]')
grid on

figure(4)
subplot(2, 2, 2)
plot(TTT, 180/pi*YYY2(:, 1));
axis([0 sampling*tend -360 +60])
xlabel('Time[s]');
ylabel('Angle of Pendulum[degree]')
grid on
%---animation---
figure(5)
xp=Lp*sin(YYY2); %pendulum xp=Lp*sin(YY(:, 2))
yp= Lp*cos(YYY2); %Lp*cos(YY(:, 2))

xa=r*sin(YYY1); %arm

hp=plot([0:0], [xp(1), yp(1)], 'LineWidth', 2, 'Marker', 'o')
%ha=plot([0:0.5], [xa(1), ya(1)], 'LineWidth', 2, 'Marker', 'o')
axis([-Lp-0.2 Lp+0.2 -Lp-0.2 Lp+0.2]);
for i=1 : size(TTT, 1)
    %set(hp, 'xdata', [0, xp(i)], 'ydata', [0, yp(i)], 'Erasemode', 'xor');
    set(hp, 'xdata', [xa(i), xa(i)+xp(i)], 'ydata', [0, yp(i)], 'Erasemode', 'xor');
    % set(hp, 'xdata', [0, xa(i)], 'ydata', [0, ya(i)], 'Erasemode', 'xor');
    drawnow
    pause(0.001);
    grid on
end
%---Animation avi---
```

```
figure(6)
xp=Lp*sin(YY2);%pendulum  xp=Lp*sin(YY(:, 2))
yp= Lp*cos(YY2);           %Lp*cos(YY(:, 2))

xa=r*sin(YY1);%arm

h1=plot([0;0], [xp(1), yp(1)], 'linewidth', 2, 'Marker', 'o');
axis([-Lp-0.2, Lp+0.2, -Lp-0.2, Lp+0.2]);
%
%

aviobj=avifile('example2.avi'); %change
for i=1 : size(TTT, 1) %TTT
    set(h1,'xdata',[xa(i), xa(i)+xp(i)], 'ydata', [0, yp(i)], 'Erasemode', 'xor');
    drawnow
    %pause(0.0001);
    grid on
    frame=getframe(gca);
    aviobj=addframe(aviobj, frame);
end
aviobj=close(aviobj);

% [Name, Path]=uiputfile({'*.avi', 'AviFiles (*.avi)'}, 'AVI-fil Save As');

%if isequal(Name, 0)||isequal(Path, 0)
%else
%    filename=fullfile(Path, Name);
%    movie2avi(aviobj, filename, 'compression', 'Cinepak');
%end
```